1-800-524-3404

# INSTALLATION INSTRUCTIONS

P/N BI 8020

OIL RESIDUE COLLECTION SYSTEM KING AIR C90A, C90GT, & C90GTI

KIT P/N BI 8030-1

03/01/11 REV: D

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## **IMPORTANT NOTICE**

THE BUTTERFIELD INDUSTRIES OIL RESIDUE COLLECTION SYSTEM INSTALLATION HAS BEEN APPROVED BY THE FEDERAL AVIATION ADMINISTRATION (FAA) UNDER SUPPLEMENT TYPE CERTIFICATE (STC).

THIS INSTALLATION MUST CONFORM TO THE FAA APPROVED BUTTERFIELD INDUSTRIES INSTALLATION DRAWINGS AND THE INSTRUCTIONS CONTAINED IN THIS DOCUMENT.

CONFORMITY VERIFICATION OF THE INSTALLED SYSTEM TO THE BUTTERFIELD INDUSTRIES INSTALLATION DRAWINGS AND INSTRUCTIONS IS THE RESPONSIBILITY OF THE INSTALLATION FACILITY.

DEVIATIONS TO THE FAA APPROVED BUTTERFIELD INDUSTRIES INSTALLATION DRAWINGS AND INSTRUCTIONS MUST BE APPROVED BY BOTH THE FAA AND BUTTERFIELD INDUSTRIES. THE INSTALLATION FACILITY IS RESPONSIBLE FOR BOTH COORDINATING AND OBTAINING THAT APPROVAL.

DRAWN		BUTTERFIELD INDUSTRIES, INC.
B. Whitcomb	05/10/00	
CHECKED		INSTALLATION INSTRUCTIONS
J. Whitcomb	05/10/00	
ENGINEER		OIL RESIDUE COLLECTION
J. Whitcomb	05/10/00	
STRESS		SYSTEM FOR THE KING AIR
APPROVED	05/15/00	MODELS C90A, C90GT & C90GTI
W. Butterfield	05/15/00	I .

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# **REVISIONS**

LTR	DESCRIPTION	APPROVALS
A	REVISION TO CHANGE DIMENSION OF TANK  REASON: TO FACILITATE A BETTER FIT INSIDE THE COWL MAKING AN EASIER INSTALLATION  PRODUCTION INFO: ALL WILL COMPLY	DRAWN: B. WHITCOMB 02/17/2001  CHECKED: J. WHITCOMB 02/17/2001
		APPROVED W. BUTTERFIELD 02/20/2001
В	REVISION TO CHANGE THE NUMBER ON REMOVAL AND INSTALLATION DRAWINGS	DRAWN: B. WHITCOMB 09/15/2001
	<b>REASON:</b> TO HAVE THE KIT PART NUMBER AND INSTALLATION DRAWING NUMBERS MATCH FOR IDENTIFICATION OF KIT	CHECKED: J. WHITCOMB 09/15/2001
	<b>PRODUCTION INFO:</b> EXISTING INSTALLATIONS SATISFACTORY, FUTURE INSTALLATIONS TO COMPLY	APPROVED W. BUTTERFIELD 09/22/2001
С	REVISION TO ADD ENGINE MODEL NUMBER	
	REASON: AIRCRAFT MANUFACTURER CHANGED ENGINE TO ADD C90GT MODEL	
	<b>PRODUCTION INFO:</b> EXISTING INSTALLATIONS SATISFACTORY, FUTURE INSTALLATIONS TO COMPLY.	CHECKED: J. WHITCOMB 10/15/06
	W. BUTTERFIELD	APPROVED: 10/23/06
DRAV	VING NUMBER BI 8020 REV. D	PAGE 3

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### **REVISIONS**

LTR DESRIPTION APPROVALS

D REVISION TO ADD MODEL TO APPLICABLITY CHECKED

W. BUTTERFILD

**REASON:** AIRCRAFT MANUFACTURER ADDED

MODEL C90GTI TO 90 SERIES LIST

**PRODUCTION INFO:** EXISTING INSTALLATION APPROVED

SATISFACTORY, FUTURE INSTALLATIONS W. BUTTERFIELD

ARE APPLICABLE FOR C90GTI

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#### 1.0 INTRODUCTION

THIS DOCUMENT IS INTENDED TO PROVIDE DETAILED INSTRUCTIONS FOR THE INSTALLATION OF THE BUTTERFIELD INDUSTRIES 'OIL RESIDUE COLLECTION SYSTEM' ON THE KING AIR C90A, C90GT & C90GTI FAMILY OF AIRCRAFT.

#### 2.0 REQUIRED DOCUMENTATION

IN ADDITION TO THIS INSTRUCTION MANUAL, THE FOLLOWING DOCUMENTS WILL BE REQUIRED TO INSTALL THE OIL RESIDUE COLLECTION SYSTEM.

#### 2.1 BUTTERFIELD SUPPLIED DOCUMENTS

THE FOLLOWING DRAWINGS OF THE CORRECT REVISION ARE SUPPLIED AS PART OF THIS DOCUMENTATION.

BI-8030 SHEET 1 AND 2 RESIDUE TANK INSTALLATION

**BI-8015 DRAWING LIST** 

#### 2.2 CUSTOMER SUPPLIED DOCUMENTS

THE FOLLOWING DOCUMENTS, OF THE LATEST REVISION IN EFFECT AT THE TIME OF THE KIT INSTALLATION, SHALL BE SUPPLIED BY THE CUSTOMER OR THE INSTALLATION FACILITY.

90-590012-13 BEECHCRAFT KING AIR C90A MAINTENANCE MANUAL

FAA-AC 43.13-1A ACCEPTED METHODS, TECHNIQUES AND

PRACTICES FOR AIRCRAFT ALTERATION.

INSPECTION AND REPAIR.

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#### 3.0 GENERAL

THE OIL RESIDUE COLLECTION SYSTEM, PART NUMBER BI-8030-1, IS ELIGIBLE FOR INSTALLATION ON THE BEECHCRAFT KING AIR C90A FAMILY OF AIRCRAFT UNDER AUTHORIZATION OF THESE INSTRUCTIONS AND THE ATTACHED STC # SA01454CH.

SERIAL NUMBER OF AUTHORIZED AIRCRAFT ARE AS FOLLOWS

LJ 1063 THRU LJ 1726 MODEL C90A OR C90B

LJ 1728 THRU LJ 1753 & LJ 1755

LJ 1727, LJ 1754 MODEL C90GT &C90GTI

LJ 1756 AND AFTER

PREMODIFIED AIRCRAFT MUST CONFORM TO INSTALLATION DRAWING BI-8030

SHEET 1 "MANIFOLD REMOVAL"

SHEET 2 "TANK INSTALL"

#### 3.1 KIT DESCRIPTION

INSTALLATION OF THE BUTTERFIELD INDUSTRIES OIL RESIDUE COLLECTION SYSTEM ALLOWS THE COLLECTING OF DRAIN FLUID INTO A SINGLE SUMP ASSEMBLY. THE SUMP HAS A DRAIN VALVE, WHICH CAN BE USED PERIODICALLY TO DRAIN THE TANK. THE SYSTEM WILL ELIMINATE THE OIL MESS ON THE BOTTOM OF THE NACELLE AND ON THE MAIN LANDING GEAR. IT WILL ALSO CLEAN UP THE MESSY OIL SPOTS ON THE HANGAR FLOOR AND RAMP WHEN THE AIRCRAFT IS PARKED. THIS IS A MESS AS WELL AS AN EPA HAZARD.

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#### 3.2 KIT CONTENTS

THE CONTENTS OF THE BI-8030-1 KIT ARE AS SPECIFIED IN THE LIST OF MATERIALS ON SHEET 1 OF THE DRAWING LISTED IN PARAGRAPH 2.1. ADDITIONALLY, THE KIT CONTAINS A DATA PACKAGE CONTAINING:

- 1) BI 8020 "INSTALLATION INSTRUCTIONS"
- 2) A COPY OF THE PACKING LIST (PAGE 12)
- 3) DRAWINGS LISTED IN PARAGRAPH 2.1
- 4) ONE COPY OF THE STC
- 5) BI 8025 INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

BUTTERFIELD INDUSTRIES SHOULD BE NOTIFIED IMMEDIATELY OF ANY SHORTAGES OR OTHER DISCREPANCIES.

#### 3.3 SPECIALIZED TOOLING REQUIRED

OTHER THAN STANDARD AIRCRAFT TECHNICIAN'S TOOLS, NO SPECIAL TOOLS ARE REQUIRED.

#### 4.0 AIRPLANE PREPARATION

NO SPECIAL AIRPLANE PREPARATION IS REQUIRED PRIOR TO INITIATING INSTALLATION OF THE OIL RESIDUE COLLECTION KIT.

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#### 5.0 REMOVALS

THE OPERATION DEFINED HERE ARE THOSE NECESSARY TO REMOVE AND ALTER COMPONENTS FROM THE LEFT AND RIGHT HAND NACELLE. THE SEQUENCE OF EVENTS IS PRESENTED IN A RECOMMENDED SEQUENCE. HOWEVER, THAT SEQUENCE MAY BE ALTERED TO SUIT THE CUSTOMER OR THE INSTALLATION FACILITY.

REF, DRAWING BI-8030 SHEET 1

- 1) GAIN ACCESS TO THE FORWARD SIDE OF THE FIREWALL BY REMOVING COWLING AS REQUIRED.
- 2) DISCONNECT HOSES FROM EXISTING DRAIN MANIFOLD.
- 3) REMOVE AND DISCARD MANIFOLD (101-910105-13) AND THE ATTACHED HARDWARE AS SHOWN.

#### 6.0 KIT INSTALLATION

THE FOLLOWING PARAGRAPHS DEFINE THE INSTALLATION OF THE OIL RESIDUE COLLECTION SYSTEM. THE ACTIVITIES SPECIFIED HEREIN ARE PRESENTED IN A RECOMMENDED SEQUENCE. HOWEVER, THAT SEQUENCE MAY BE ALTERED TO SUIT THE CUSTOMER OR THE INSTALLATION FACILITY. UNLESS OTHERWISE SPECIFIED, INSTALLATION OF ALL HARDWARE SHALL CONFORM TO FAA-AC 43.13-1A.

#### 6.1 SUMP ASSEMBLY INSTALLATION

REF: DRAWING BI 8030 SHEET 2

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1) POSITION SUMP ASSEMBLY (BI 8005-1) ON THE ENGINE FIREWALL AS SHOWN.

NOTE: THE EXISTING HOLE IN THE ENGINE COWL MAY REQUIRE ENLARGMENT TO ACCEPT THE NECK OF THE SUMP ASSEMBLY.

- 2) IT WILL BE NECESSARY TO DRILL A .500 HOLE IN COWLING TO ACCOMMODATE OVERFLOW TUBE.
- 3) ATTACH THE SUMP ASSEMBLY WITH TWO BOLTS (AN3H3A) AND WASHERS (AN960KD10) AT TWO EXSITING NUTPLATE LOCATIONS.
- 4) SAFETY-WIRE THE TWO BOLTS AS SHOWN.

#### 6.2 HOSE INSTALLATION

CONNECT HOSES TO SUMP ASSEMBLY IN SAME POSITIONS.

#### 6.3 FINAL INSTALLATION

- 1) PERFORM A DETAILED INSPECTION OF ALL WORK ACCOMPLISHED
- 2) INSTALL ALL COWLING REMOVED TO GAIN ACCESS FOR THE MODIFICATION.

NOTE: DRAIN VALVE MAY BE REMOVED DURING COWLING INSTALLATION, AND THEN REINSTALLED.

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#### 6.4 INSPECTION

THERE IS NO INSPECTION CRITERIA FOR THE SUMP. CHECK CONTINUED AIRWORTHINESS INSTRUCTIONS ON PAGE 13 FOR SUGGESTIONS AS TO PERIODIC MAINTENANCE.

#### 7.0 RETURN AIRPLANE TO SERVICE

1) COMPLETE AN APPROPRIATE ENTRY IN THE AIRPLANE LOGBOOK

EXAMPLE: INSTALLED BUTTERFIELD INDUSTRIES OIL

RESIDUE COLLECTION SYSTEM PART

NUMBER BI 8030-1 AS PER STC SA01454CH.

- 2) COMPLETE THREE COPIES OF FAA FORM 337. INSERT ORIGINAL IN AIRPLANE FLIGHT MANUAL. MAIL ONE COPY TO LOCAL FAA-FSDO AND RETAIN ONE COPY FOR THE INSTALLER
- 3) NO WEIGHT AND BALANCE CHANGE

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# **PACKING LIST**

### KIT P/N BI 8030-1

RESIDUE TANK ASSEMBLY	BI 8005-1	2 EA.
WASHER	AN 960KD10	4 EA.
ROLT	AN3H3A	4 F A

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#### INSTRUCTIONS FOR CONTINUED AIRWOTHINESS

THERE ARE NO SCHEDULED INSPECTION PERIODS FOR ANY COMPONENT OF THE BUTTERFIELD INDUSTRIES 'OIL RESIDUE COLLECTION SYSTEM'.

USE THE FOLLOWING GUIDELINES FOR SUCCESSFUL CONTINUED AIRWORTHINESS.

ANT TIME THE COWLING IS OPEN FOR INSPECTION, THE OIL RESIDUE TANK SHOULD BE INSPECTED FOR CONDITION AND SECURITY.

ALL HOSE CLAMPS TO THE TANK SHOULD BE INSPECTED FOR CONDITION AND SECURITY.

THE TANK SHOULD BE DRAINED ON A PERIODIC BASIS.

IF THE TANK IS NOT DRAINED, THE RESIDUE OIL WILL DRAIN FROM THE OVERFLOW TUBE.

03/01/11 REV: N/C



